SPACE HAUC is a 3U CubeSat mission funded by NASA, designed and tested entirely by undergraduates at UMass Lowell.

The SPACE HAUC satellite’s purpose is to transmit high frequency data using an X Band Phased Array antenna.

Satellites that rely on transmitting information down to earth have to be pointed at the ground. SPACE HAUC uses solar panels, and has to be pointed at the sun.

Attitude is the orientation of the satellite, thus attitude control is crucial for the utility and lifetime of a satellite, and often involves attitude determination, or the tracking of the satellite’s current orientation.

Attiude control is needed to correct disturbance torques from the atmosphere, the gravity gradient, the structure’s magnetic dipole, and solar radiation.

We control spin the spacecraft for stability and control using magnetic torquers.

SPACE HAUC’s Attitude Control System is comprised of multiple physical components: a magnetometer, several “coarse sun sensors”, a “fine” sun sensor, three magnetorquers.

We only set voltage through magnetorquers at certain values (-5 V, 0 V, 5V).

Detumbling: Reduce spinning around X,Y axes. Spin satellite around Z (perpendicular to camera, panels).

Due to time and budget constraints, we do not calculate absolute attitude.

Sun Pointing: Determine where the sun is using sensors, torque to that attitude.

Gyroscopes are not able to be tested in the lab, so we use the magnetic field values from the magnetometer to determine angular velocity. We use a Kalman filter to do this.

We use Simulink to calculate the orbit, rotation, magnetic field values, and sun and earth vectors.